| MAJOR REPAIR AND ALTERATION (S. Department Transportation (Airframe, Powerplant, Propeller, or Appliance) | | | | | OMB No.2120-0020 2/28/2011 For | FAA Use Only | |
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| J.S. Department of Transportation deral Aviation ministration | | | | | 3.9-1 (or subseque | nt revision thereof) for | |
| lou doddono di | lation (49 U.S.C. | §46301(a)). | ee Title CFR §43.9, Part 4 report is required by law (| 49 U.S.C. §44701). Fai | ure to report can re | sult in a civil perialty is | |
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| 2. Owner | Name (As shown on registration certificate) DAN A HUGHES MANAGEMENT LLC LOS MEDIO CORP | | | Address 208 E City BEEV | | State TEXAS | |
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AIRFRAME

Certificate or Designation No.

(10-06)

Q2AR598N

FAA Form 337

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| | | | 6. | Confo | rmity State | ment | | |
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| Name | ADVANTAGE All | RCRAFT SER | VICES INC. | | U.S. Certifica | ated Mechanic | | Manufacturer |
| Address | 355 PINSON DR | | | | Foreign Cert | ificated Mechanic | | C. Certificate No. |
| City | CORPUS CHRIS | TI | State TEXAS | _ X | Certificated I | Repair Station | | Q2AR598N CL 3 A/F |
| Zip | 78406 | _Country _L | ISA | | Certificated I | Maintenance Organizat | tion | |
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| BY | FAA Flt. Stand Inspector | dards | Manufacturer | | Maintenar | ce Organization | | Person Appoved by Canadian Department of Transport |
| | FAA Designe | ee X | Repair Station | | Inspection | Authorization | Other | (Specify) |

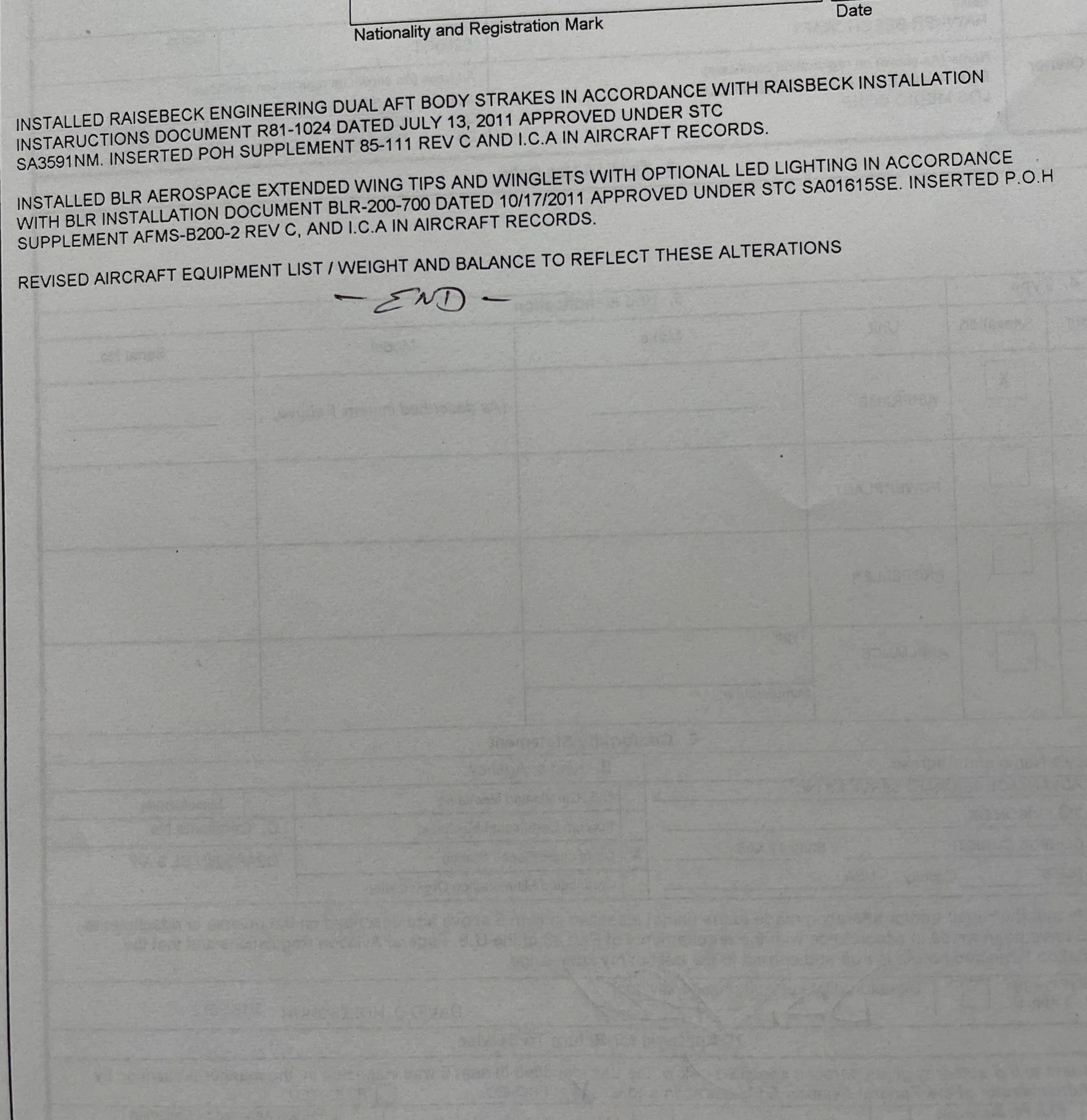
Signature/Date of Authorized Individual

DAVD D. HOLZSCHUH 3/12/2012

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Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

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| 8. Description of Work Accor | plished each additional sheets. Identify with aircraft nationality and regis | stration mark and date work completed.) |
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| | Nationality and Registration Mark | Date |



Department of Transportation—Federal Aviation Administration

Supplemental Type Certificate

Number SA3591NM

This certificate, issued to

Raisbeck Engineering 4411 South Ryan Way Seattle, WA 98178

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part 23 of the Federal Aviation Regulations.

Original Product—Type Certificate Number:

A24CE

Make:

Beech

Model:

200, 200C, 200T, 200CT, A200, A200C, A200CT,

for

B200, B200C, B200T, B200CT, B200GT,

B200CGT

Description of the Type Design Change: Installation of lower aft fuselage strakes in accordance with FAA-Approved Raisbeck Master Drawing List R81-1004 Revision H, dated December 5, 2007, or later FAA-approved revision.

Limitations and Conditions: Approval of this change in type design applied to the above model aircraft only. This approval should not be extended to other aircraft of this model on which other previously approved modifications have been incorporated unless it is determined that the relationship between this change and any of those other previously approved modifications, including changes in type design will introduce no adverse effect upon the airworthiness of that aircraft. A copy of this certificate and FAA-approved Aircraft Flight Manual Supplement (AFMS) 85-111 Rev. C, dated December 5, 2007, shall be maintained as part of the permanent records for the modified aircraft.

If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application:

May 12, 1991

Date reissued:

Date of issuance:

January 15, 1992

Date amended:

By direction of the Administrator

February 9, 1994, December 2, 2005,

December 7, 2007, March 31, 2011

(Signature)

Acting Manager, Seattle Aircraft Certification Office

(Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or Imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.

FAA FORM 8110-2(10-68)



| | | | REV | ADCN | DATE |
|----------|-----|--|-----|------|----------|
| IG | SHT | TITLE | | | 7/13/11 |
| R81-1024 | | Installation Instructions - Raisbeck Aft Body Strake Installation | H | | 9/8/99 |
| 91 2402 | 1 | Aft Body Strake Assembly | D | 2 | 9/0/99 |
| 81-3402 | | | E | 2 | 7/17/06 |
| 81-3405 | 1 | Aft Body Strake Details | В | 1 | 2/1/88 |
| 81-3405 | 2 : | Aft Body Strake Details | C | | 10/16/97 |
| 81-3405 | 3 | Aft Body Strake Details | G | | 4/23/01 |
| 81-3405 | 4 | Aft Body Strake Details | A | | 9/8/99 |
| 81-3405 | 5 | Aft Body Strake Details | | | 11/9/82 |
| 81-3414 | 1 | Ventral Fin Plug | IR | | 11/7/02 |
| 01-3414 | | | R | 2 | 06/15/10 |
| 200-5302 | PL | Strake Installation – Factory Option | В | 1 | 9/24/09 |
| 200-5302 | 1 | Strake Installation – Factory Option | | | |
| 85-111 | | FAA-Approved Pilots Operating Handbook Supplement & Aircraft Flight Manual For The Beechcraft Super King Air 200/200C/200T/200CT/A200/A200CT/B200/ B200C/B200T/B200CT/B200GT/B200CGT Equipped With Raisbeck Dual Aft Body Strakes | C | | 12/5/07 |
| RES-102 | | Process Specification – Installation of Permanent Fasteners | A | | 4/11/83 |
| RES-103 | | Process Specification – Part Markings | В | | 4/4/83 |

MASTER DRAWING LIST

Russell Bezzo APPROVAL DESIGNEE

| DRAWING NO. R81-1004 | STC SA359 | 91NM | REV LVL |
|-----------------------------------|-----------------|------|---------|
| DRAWING LIST | | | |
| Raisbeck Dual Aft Body Stake | Installation | | M |
| Beechcraft Super King Air 200/200 | OC/200T/200CT/A | 200/ | IAI |
| A200CT/B200/B200C/B200T/B20 | | | |
| Approval Date: 7/13/ | | PG | 1 of 1 |

Department of Transportation Federal Aviation Administration

Supplemental Type Certificate

Number SA01615SE

This certificate, issued to:

BLR Aerospace, LLC 3102 100th Street SW Everett, WA 98204

certifies that the change in the type design for the following product with the limitations and conditions therefore as specified hereon meets the airworthiness requirements of Part 23 of the Federal Aviation Regulations.

Original Product - Type Certificate Number:

A24CE

Raytheon

200, 200C, B200, B200C, B200GT, B200CGT, 300, and 300LW

Description of the Type Design Change: Installation of winglets and associated changes in wing and aileron structures and related system changes on Raytheon Model 200, 200C, B200, B200C, B200GT B200CGT, 300, and 300LW airplanes. Fabrication and installation of parts must be accomplished in accordance with FAA-approved Master Drawing List KAW190040, Revision N/dated February 13, 2008, or later FAA-approved revision/

Limitations and Conditions: The limitations and conditions of Type Certificate A24QE apply except as outlined on this STC. Compatibility of this design change with previously approved modifications must be determined by the installer.

A copy of this Certificate, the FAA-approved Airplane Flight Manual Supplement, and Instructions for Continued Airworthiness must be maintained as part of the permanent record of the modified airplane. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written permission.

(See Continuation Sheets on Pages 3 and 4)...

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Avlation Administration.

Date of application: Date of issuance:

June 06, 2003

Date reissued:

November 23, 2009

November 02, 2005

Date amended: October 13, 2006, April 02, 2007,

July 29, 2008

By direction of the Administrator

(Signature)

Acting Manager, Seattle Aircraft Certification Office

(Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.

FAA FORM 8110-2-1 (10-69)

PAGE 1 OF 4 PAGES

Department of Transportation—Federal Aviation Administration

Supplemental Type Certificate

(Continuation Sheet)

Number SA01615SE

BLR Aerospace, LLC

Issued:

November 02, 2005

Reissued:

November 23, 2009

Amended:

October 13, 2006, April 02, 2007, July 29, 2008

Limitations and Conditions continued:

Airworthiness & Environmental Standards for components and areas not affected by the change are the original certification basis for the Models 200, 200Ç, B200, B200C, B200GT, B200CGT, \$000, and \$00LW as shown on TCDS A24CE Revision 97. Airworthiness and #nvironmental Standards for components and areas affected by the change are as specified in the following HAA-approved Master Drawing List.

KAW190040, Revision N, dated February 13, 2008, or later FAA-approved revision.

Eligible Serial Numbers:

Eligible serial numbers are listed in the following FAA approved Master Drawing List:

KAW190040, Revision N, dated February 13, 2008, or later FAA-approved revision.

Instructions for Continued Airworthiness:

Airplanes modified in accordance with this STC must be maintained in accordance with the applicable Instructions for Continued Airwarthiness document specified below:

BLR-200-950, Revision F, dated June 5, 2007, dr later FAA-approved revision for Models 200, 200C, B200, B200C, B200G/T, and B200CGT.

BUR-300-950, Revision C, dated Juhe 5, 2007, or later FAA-approved revision for Models 300 and 300LW.

For aircraft without Gurney Flap installed:

Structural life limits and other requirements for the wing structure are changed by the installation of this STC. Specifically, the structural life limits for the wing structure, the accomplished times specified for initial and recurring wing structural inspections and the compliance times for Airworthiness Directives affecting Raytheon Model 200, 200C, B200, B200C, B200GT, B200CGT, 300, and 300LW wing structure, as expressed in flight hours may be reduced by up to 35% by the installation of this STC. The means and requirement to establish the life limits and required compliance times are provided in the applicable Instructions for Continued Airworthiness document.

(See Continuation Sheet on Page 4)...

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.

FAA FORM 8110-2-1 (10-69)

PAGE 3 OF 4 PAGES

Department of Transportation—Federal Aviation Administration

Supplemental Type Certificate

(Continuation Sheet)

Number SA01615SE

BLR Aerospace, LLC

Issued:

November 02, 2005

Reissued: November 23, 2009

Amended: October 13, 2006, April 02, 2007, July 29, 2008

Limitations and Conditions continued:

Airplanes modified in accordance with this STC must be operated in accordance with the applicable Airplane Flight Manual Supplement specified below.

AFMS-B200-1, Revision IR, dated October 17, 2005, or later FAA-approved revision, for Models 200, 200C, B200, and B200C.

AFMS-B200-2, Revision B, dated December 21, 2007, or later FAA-approved revision, for Models 200, 200C, B200, B200C, B200GT, and B200CGT equipped with optional de-ice boots and/or Gurney flaps.

AFMS-300-1, Revision IR dated October 11, 2006, or later FAA-approved revision, for Models 300 and 300LW

AFMS-300-2, Revision A, dated June 01, 2007, or later FAA-approved revision, for Models 300 and 300LW.

Eligible Autopilot Systems:

Autopilot systems that have been evaluated for compatibility with this STC are identified in the following Master Drawing List:

KAW190040, Revision N, dated February 13, 2008, or later FAA-approved revision.

- END -

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.

FAA FORM 8110-2-1 (10-69)

PAGE 4 OF 4 PAGES



BLR Aerospace, LLC 3102 100th Street SW Everett, WA 98204 (US & Canada) 800.257.4847 (International) 425.353.6591 (FAX) 425.355.3046 www.BLRaerospace.com

STC License

BLR Aerospace, LLC Serial Number Specific STC'd Products

BLR Aerospace, LLC, grants a 'one time' license to modify this aircraft (identified by serial number) in accordance with the enclosed STC. The installation of components in accordance with the STC Installation Manual is only authorized on the serialized humbered aircraft that appears on this letter and the copy of the STC provided with this kit.

The serialized copy of the STC, and this letter, are to be kept as part of the permanent records for the modified aircraft.

Instructions to Installer

The STC copy provided with this kit, and the Airplane Flight Manual Supplement (if required) should be checked for proper serial and registration numbers. The copy of the STC will have the aircraft serial number across each page (corner to corner). The 'serialization' of the STC copy is the method by which BLR Aerospace, LLC, authorizes the installation of its product.

Should the serial or registration number not reflect that of the aircraft that is being modified, BLR Aerospace, LLC, is to be notified immediately by telephone (425-353-6591), or by fax (425-355-3046), or email (mike@BLRaerospace.com). The aircraft should not be modified until BLR Aerospace, LLC, has been notified, and any documentation errors have been corrected.

Mike Carpenter

President BLR Aerospace, LLC

Performance Innovation

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| US Departmen | t |
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MAJOR REPAIR AND ALTERATION

Electronic Tracking Number Form Approved OMB No. 2120-0020 2/28/2011 For FAA Use Only

(Airframe, Powerplant, Propeller, or Appliance) INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and discount revision than the second discount revision that the second discount revision that the second discount revision than the second discount revision that the second discount revision than the second discount revision that the second discount revision than the second discount revision that the second discount revision that the second discount revision that instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a)) Serial No. Nationality and Registration Mark BY-30 Series N94CK Model 1. Aircraft SUPER KING AIR Make B200GT Address (As shown on registration certificate) HAWKER BEECHCRAFT CORP Address 286 GRADY MONTGOMERY DR Name (As shown on registration certificate) State TN JACKSON Country US 2. Owner 38301-9603 Zip KIRKLAND AVIATION INC 3. For FAA Use Only 5. Unit Identification Serial No. 4. Type Model Make Unit Alteration Repair (As described in Item 1 above) AIRFRAME 1 POWERPLANT PROPELLER Type APPLIANCE Manufacturer 6. Conformity Statement B. Kind of Agency A. Agency's Name and Address Manufacturer U. S. Certificated Mechanic C. Certificate No. Name STEVENS AVIATION Foreign Certificated Mechanic Address 707 HANGAR LANE Certificated Repair Station State TN NASHVILLE VIB3368K Certificated Maintenance Organization Country US D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto Zip have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge. Signature/Date of Authorized Individual Extended range fuel per 14 CFR Part 43 8-4-2009 App. B 7. Approval for Return to Service Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is Persons Approved by Canadian Maintenance Organization Department of Transport FAA Flt. Standards Manufacturer Other (Specify) Inspector Inspection Authorization BY Repair Station FAA Designee Signature/Date of Authorized Individual

VIB3368K FAA Form 337 (10-06) 8-4-2009

Certificate or

Designation No.

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N94CK

Nationality and Registration Mark

Date

11 199

Installed Hawker Beechcraft RVSM STC SA011798SE iaw Hawker Beechcraft Instruction for Initial and

Continued Airworthiness doc AMI-IICA-KAIR-0714, dated 05/01/08. Inserted FAA Approved Aeromech Flight Manual Supplement doc AMI-AFMS-KAIR-0719 dated 11-21-07

into the Pilot's Operating Handbook. See the attached ICA document AMI-IICA-KAIR-0712, DATED 5-1-08

No change to W&B

A log entry showing this alteration has been made.

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Form Approved
OMB No. 2120-0020

Electronic Tracking Number

2/28/2011 For FAA Use Only MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance) US Department of Transportation INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition. **Federal Aviation** instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (40 U.S.C. \$4000). such violation. (49 U.S.C. §46301(a)) Serial No. Nationality and Registration Mark BY-30 N94CK Model . Aircraft SUPER KING AIR Make **B200GT** HAWKER BEECHCRAFT CORP Address (As shown on registration certificate) Name (As shown on registration certificate) Address 286 GRADY MONTGOMERY DR State TN JACKSON City 2. Owner Country US 38301-9603 KIRKLAND AVIATION INC 3. For FAA Use Only 5. Unit Identification 4. Type Serial No. Model Make Unit Alteration Repair (As described in Item 1 above) AIRFRAME POWERPLANT PROPELLER Type APPLIANCE Manufacturer 6. Conformity Statement B. Kind of Agency A. Agency's Name and Address U. S. Certificated Mechanic Manufacturer STEVENS AVIATION Address 707 HANGAR LANE C. Certificate No. Foreign Certificated Mechanic Certificated Repair Station State TN NASHVILLE City Country US Certificated Maintenance Organization VIB3368K 37217 Zip I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge. Signature/Date of Authorized Individual Extended range fuel per 14 CFR Part 43 8-4-2009 App. B 7. Approval for Return to Service Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is Rejected Approved FAA Flt. Standards Persons Approved by Canadian Maintenance Organization Manufacturer Department of Transport Inspector BY Other (Specify) Inspection Authorization Repair Station FAA Designee Signature/Date of Authorized Individual Certificate or Designation No.

VIB3368K FAA Form 337 (10-06) 8-4-2009

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

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Certificate or Designation No.

VIB3368K

FAA Form 337 (10-06)

FAA Flt. Standards

Inspector

FAA Designee

8-4-2009

Manufacturer

Repair Station

Signature/Date of Authorized Individual

Maintenance Organization

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Persons Approved by Canadian

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Other (Specify)

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Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

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VENDOR CAGE CODE 70898
HAWKER BEECHCRAFT CORP
RLSD 12/05/07
DOC0000325514 R1 V1

Whited States of America

Department of Transportation — Federal Aviation Administration

Supplemental Type Certificate

Number SA01798SE

This certificate, issued to

Hawker Beechcraft Corporation 9709 E. Central Wichita, KS 67201-0085

certifies that the change in the type design for the following product with the limitations and conditions therefore as specified hereon meets the airworthiness requirements of Part 23 of the Federal Aviation Regulations.

Original Product - Type Certificate Number:

A24CE (Revision 96)

Make: Model: Hawker Beechcraft B200, B200C, B200GT, B200CGT, for

ach

B300, B300C

Description of the Type Design. Change: Group approval for Reduced Vertical Senaration Minimum (RVSM) in accordance with AeroMech Document No. AMI-QD-KAIR-0710, Rev. "NEW," dated November 16, 2007, or later FAA approved revisions.

Limitations and Conditions: Instructions for Initial and Continued Airworthiness, Document AMI-IICA-KAIR-0714, Rev. "New," dated November 16, 2007, or later approved revision, Is a required part of this STC.

The Airplane Flight Manual Supplement for the Hawker Beechcraft models B200, B200C, B200GT, and B200CGT, AeroMech Document AMI-AFMS-KAIR-0719, Rev. "Original", FAA Approved November 21, 2007, (or latest FAA-approved revision), is required on-board for operation of the aircraft.

The Airplane Flight Manual Supplement for the Hawker Beechcraft models 8300 and 8300C, AeroMech Document AMI-AFMS-KAIR-0722, Rev. "Original", FAA Approved November 21, 2007, (or latest FAA-approved revision), is required on-board for operation of the aircraft.

Descriptive data pertaining to this design change are considered inadequate for duplication on other products. This approval is limited to the Hawker Beechcraft Models B200, B200C, B200GT, B200CGT, B300, B300C aircraft only.

A copy of this certificate and FAA approved flight manual supplement must be maintained as part of the permanent records of the aircraft. If the holder agrees to permit another person to use this certificate, the holder shall give the other person written evidence of the permission.

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application:

August 20, 2007

Date of issuance:

November 21, 2007

Date reissued.

Date amended.



By direction of the Administrator

(Signature)

Acting Manager, Seattle Aircraft Certification Office

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.

Fourter Beechcraft

9709 E. Central Wichita, KS 67201-0085 Web: https://www.hawkerbeechcraft.com/

for

each

Instructions for Initial and Continued Airworthiness for
Hawker Beechcraft Model
B200/B200C/B200GT/B200CGT/B300/B300C
Airplanes Qualified for Operations within
Reduced Vertical Separation Minimum (RVSM) Airspace

Supplemental Type Certificate No: SA01798SE

| Airplane Model | Airplane Serial Number(s) |
|----------------|--|
| B200 | The Theory of A.A. T. Confidence of the second of the seco |
| B200C | |
| B200GT | the property of the property o |
| B200CGT | |
| B300 | |
| B300C | The Lockett arend exil month in the 2 left for |

The inspections and tests required for RVSM initial airworthiness compliance must be conducted per the instructions in this document, prior to receiving RVSM operational approval. After RVSM approval is granted, the operator must perform inspections and tests required for continued airworthiness compliance, also detailed in this document.

These instructions must be added to the existing maintenance and flight operations programs for the airplane to conduct operations within RVSM airspace. Additional care, calibration, and monitoring of the airplane's air data and automatic altitude control systems (with special attention in the neighborhood of the static pressure ports) must be taken. Any deviation from these procedures must be coordinated through Hawker Beechcraft Corporation (herein Hawker Beechcraft) and the responsible airworthiness authority prior to operation within RVSM airspace.

Revisions to this document must be coordinated and approved through the Supplemental Type Certificate (STC) Holder. The STC Holder will track all changes in the Log of Revisions and identify the revision level of every page in the List of Effective Pages. All revisions to this document will be redistributed (by the STC Holder) to the owner of the applicable airframe for incorporation.

The information presented in this document supplements or supersedes the basic manual only in those areas specified. For maintenance procedures pertaining to the airplane's systems and specific avionics equipment not covered in these instructions, see the manufacturer's requirements and procedures.

STC Holder:

Hawker Beechcraft Corporation

9709 E. Central

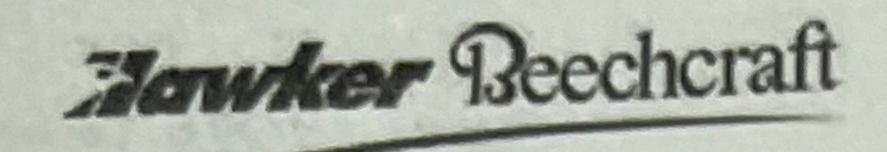
Wichita, Kansas 67201-0085

RLSD 05/01/08

AMI-IICA-KAIR-0714

REV 3

Page i



RVSM Approval for Hawker Beechcraft B200/B200C/B200GT/B200CGT/B300/B300C Instructions for Initial and Continued Airworthiness Supplemental Type Certificate #SA01798SE

Log of Revisions

This document must be maintained current. Each time a change is required the STC Holder will track all changes in the Log of Revisions and the revision level of every page in the List of Effective Pages. Paper copies of the revisions will be redistributed via courier (by the STC Holder) to all owners of STC #SA01798SE. Upon receipt of the revision, the revised pages must be inserted into this document and all superseded pages disposed of.

It is the responsibility of the operator to ensure that this document is current prior to having any initial or continued airworthiness maintenance performed. Contact the STC Holder for verification of the current revision level.

| Revision | Revision Pages Description | | Date | Approval | | |
|--|--------------------------------------|--|-------------|----------------------------|--|--|
| NEW | NEW i - v, 1 - 25 Initial Release | | 16 Nov 2007 | ady C. Whin | | |
| Removed "FAA Approv Statement for Revision 1— change to Limitation Sectors REV 2 i—v, Revised the Model No. and | | Removed "FAA Approved" Statement for Revision 1 – No change to Limitation Section. Revised the Model No. and Part No. of DCP 1 and DCP 2 in Table 1 on page 5. | 14 Dec 2007 | | | |
| REV 3 | i-v, 1-25 | This document has been revised in its entirety to revision 3. Added footnote and asterisk to page 2 table 1 for models B200 and B200C. | 9 Jan 2008 | | | |
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OMB No. 2120-0020 Electronic Tracking Number Exp: 5/31/2018 For FAA Use Only

| nstructions | n. (49 U.S.C | C. §46301(a)) | | | C. §44701). Failure to repo | rt can res | sequent revision thereof) for ult in a civil penalty for each |
|--|--|--|--|--|--|-------------------------------------|--|
| I. Aircraft | US Regi | and Registration stered N322 | | | | Ts | eries o |
| | Make Beechcraft | | | | Model B200GT | | B200 |
| | | | Address (As shown on registration certificate) | | | | |
| | Name (As shown on registration certificate) DAN A HUGHES MANAGEMEN | | | LLC Address 208 E HOUSTON ST City Beeville Address 208 E HOUSTON ST State TX | | | State TX |
| Owner | DAN A HUGHES MANAGEMENT | | | City Beeville Zip 78102-4820 Country US | | | ry US |
| | | | | | | | |
| 4. Ty | pe | | | 5. Unit Identifica | | | Serial No. |
| Repair | Alteration | Unit | Mak | е | Model | - h | BY30 |
| | × | AIRFRAME | Beechcraft | | (As described in Item 1 | above) | |
| | | POWERPLANT | | | | | |
| | | PROPELLER | | | | | |
| | | APPLIANCE | Manufacturer | | | | |
| | | | 6. | Conformity Sta | | F # 340 * \$10 | PARTIES AND THE PROPERTY OF THE PARTY. |
| | | | | B. Kind of Agen | | I Ma | anufacturer |
| A. Agency's | Name and A | Address | | I II S Certific | cated Mechanic | | |
| Name Textr | on Aviation San A | | | | tificated Mechanic | | tificate No. |
| Name Textr | | | State TX | Foreign Cer Certificated | tificated Mechanic Repair Station | | |
| Name Textra Address 1758 Sk City San 7 | on Aviation San A syplace BLVD Antonio | Antonio USA | | Foreign Cer Certificated Certificated | tificated Mechanic Repair Station Maintenance Organization | C. Cer | CNQ5918C |
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| Name Textra Address 1758 Sk City San / Zip 7821 D. I certif have furnis Extended raper 14 CFR App. B | Antonio fy that the replace made in hed herein is ange fuel Part 43 | pair and/or alterate accordance with true and correct | ion made to the unit(so the requirements of to the best of my known ature/Date of Authorized 7. Ap. | Foreign Cer Certificated Certificated identified in iter Part 43 of the U. Medge. zed Individual proval for Retu | Repair Station Maintenance Organization m 5 above and described or S. Federal Aviation Regulation rn to Service | the reversions and to | CNQ5918C rse or attachments hereto hat the information |
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| Name Textra Address 1758 Sk San A Textra Tex | Antonio fy that the replace made in hed herein is ange fuel Part 43 | pair and/or alterate accordance with true and correct Sign | ion made to the unit(so the requirements of to the best of my known ature/Date of Authorizations specified below | Foreign Certificated Certificat | rn to Service tificated Mechanic Repair Station Maintenance Organization m 5 above and described or S. Federal Aviation Regulation rn to Service tified in item 5 was inspersed Reje | cted in cted | CNQ5918C rse or attachments hereto hat the information |

FAA Form 337 (10/06)

Page 1

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

| 8. Description of Work Accomplished (If more space is required, attach additional sheets. Identify with aircraft re | nationality and registration mark and | date work completed.) |
|--|---|-------------------------------|
| | US N322DH | May 18 2018 |
| | Nationality and Registration Mark | k Date |
| Rockwell Collins Mode | ernization Upgrade | |
| Modified existing Rockwell Collins Proline 21 Systement of the Collins Proline 21 System of the Col | m to add Synthetic Visio | n, WAAS, ADS-B and |
| Complied with STC SA01434WI-D for the modification GPS-4000S WAAS receiver Global Positioning Sys | ion of Rockwell Collins Cotem. Inserted AFMS AC | SPS-4000A to C-07-009 in AFM. |
| Complied with STC SA01848WI for the modification WAAS receiver Global Positioning System. This ST substitution and procedure for installation of a Rock existing Rockwell Collins GPS-4000S receiver in 14 Revision - December 12, 2016 in log books. | C provides the approved well Collins GPS-4000S | receiver in place of an |
| Modified existing Rockwell Collins File Server Unit i | n accordance with STC | SA01676WI. |
| Complied with STC SA01430WI-D for the modificat DBU-5010E. | ion of DBU-4100 Data B | ase Unit to a |
| Complied with STC SA10965SC for FMS-3000 LPV into the Aircraft Flight Manual. Inserted ICA 324-00- | | |
| Complied with installation of ADS-B Out and Synthese SA11133SC. Inserted AFMS 543-00-0055 REV. E is 543-00-0056 Revision: B October 18, 2016 in maintenance. | into the AFM. Inserted IC | |
| Performed EMI/RFI system checks of existing system effects noted. | ems with new installation | above, no adverse |
| Updated Flight Manual Weight and Balance to inclu | ide the Equipment List. | |
| FNI | | |
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| Additional She | eets Are Attached | |

FAA Form 337 (10/06)

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